



# ME 323: FLUID MECHANICS-II

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**Lecture-11**

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**Blow Down Process**

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## Problem

An air tank of volume  $1.5 \text{ m}^3$  is at  $800 \text{ kPa}$  and  $20^\circ\text{C}$  when it begins exhausting through a converging nozzle to sea-level condition  $p = 100 \text{ kPa}$ . The throat area is  $0.75 \text{ cm}^2$ . Estimate:

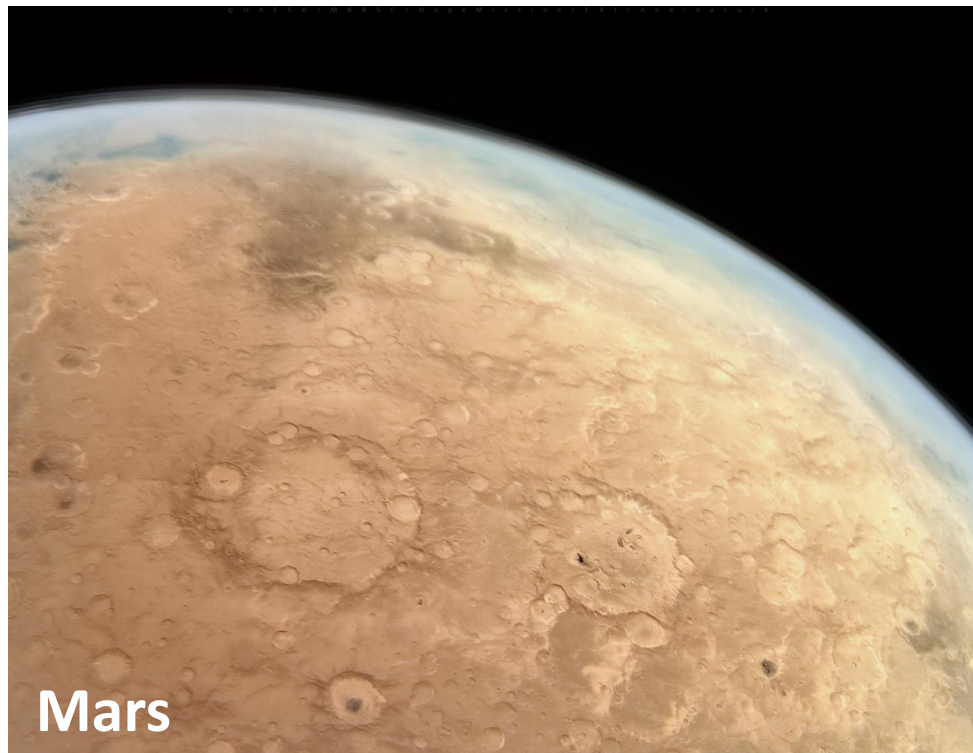
- 1) Initial mass flow rate
- 2) Time to blowdown to  $500 \text{ kPa}$  and mass flow rate at this condition
- 3) Time when the nozzle ceases being choked and mass flow rate at this conditions
- 4) Graphically show the unsteady process until the nozzle is choked.

$$t \approx 1.75 \frac{V}{a_0 A_e} \ln \left[ \frac{p_{0,0}}{p_{0,t}} \right]$$



# Fluid dynamics to explore ... ..

## Beyond the earth, toward Mars



The atmosphere of Mars is the layer of gases surrounding Mars. It is primarily composed of **Carbon dioxide (95%)**, nitrogen (2.85%), and argon (2%). It also contains trace levels of water vapor, oxygen, carbon monoxide, hydrogen, and noble gases.

TABLE I. Characteristics of Martian and Earth's atmospheric environment.

Features	Mars	Earth
Acceleration of gravity ( $\text{m/s}^2$ )	3.72	9.78
Atmospheric pressure (Pa)	640	101 300
Air density ( $\text{kg/m}^3$ )	0.0167	1.22
Mean temperature ( $^{\circ}\text{C}$ )	-63	15
Sound velocity (m/s)	227	340
Atmospheric dynamic viscosity [ $\text{kg}/(\text{m s})$ ]	$1.289 \times 10^{-5}$	$1.789 \times 10^{-5}$
Atmospheric constants (J/kg/K)	188	287
Specific heat capacity ratio( $\gamma$ )	1.29	1.40
Molar mass (g/mol)	44.01	28.96

### Challenges in designing air vehicle in Mars:

- Low Re
- Compressibility, rarefaction
- $C_L, C_D$  ???

